

AIRPLANE FLIGHT MANUAL

MODEL S-1T AIRPLANE



FAA APPROVED:_

CHIEF, ENGINEERING AND MANUFACTURING BRANCH, SOUTHERN REGION, FEDERAL AVIATION AGENCY

DATE:______July 24, 1975

PITTS AEROBATICS AIRPLANE FLIGHT MANUAL

MODEL S-1T AIRPLANE

LOG OF REVISIONS

REVISION LETTER	PAGES AFFECTED	DESCRIPTION OF CHANGE	APPROVAL AND DATE
А	i, 1 of 10, 10.1 of 10	typo., and placards added	Acting Chief, Engineering and Manufacturing Branch Southern Region, FAA Date: October 29, 1976
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PITTS AEROBATICS AIRPLANE FLIGHT MANUAL MODEL S-1T AIRPLANE

A. <u>Airspeeds:</u>

SECTION I OPERATING LIMITATIONS

Normal operating range (green arc) from 64 MPH stall speed: 56 KNOTS
To maximum normal operating speed: 154 MPH 134 KNOTS
Caution range (yellow arc) from maximum 154 MPH structural cruise speed: 134 KNOTS
To never exceed speed: (red radial) 203 MPH 176 KNOTS
NOTE: DO NOT OPEN CANOPY PAST FIRST NOTCH ABOVE 120 MPH (104 KNOTS)
FOR ACROBATIC MANEUVER ENTRY SPEEDS, SEE PLACARDS SECTION.
B. Powerplant Limits:
For Lycoming AEIO-360-AlE engine and Hartzell HC-C2YK-4CF/FC7666A-2 propeller. Propeller minimum diameter 72 inches. Propeller maximum diameter is 74 inches.
Propeller Pitch Settings: High Pitch $28^{\circ} \pm \frac{1}{2}^{\circ}$
Low Pitch 13½°
Engine rated power: 200 HP @ 2700 RPM
Minimum fuel grade: 100 Octane
Oil Pressure:
Minimum (red radial) 25 PSI
Caution range (yellow arc) from 25 PSI to 60 PSI
Normal range (green arc) from 60 PSI to 90 PSI
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SECTION 1

OPERATING LIMITATIONS

B. Powerplant Limits (cont'd))
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B. Powerplant Limits (cont'd)		
Oil Pressure: (cont'd)		
Caution range (yellow arc)	from to	90 PSI 100 PSI
Maximum (red radial)		100 PSI
Oil Temperature:		
Maximum (red radial)		245 Deg. F 118 Deg. C
Normal range (green arc)	from	100 Deg. F
	to	38 Deg. C 245 Deg. F 118 Deg. C
Fuel Pressure:		
Minimum		O PSI
Normal range (green arc)	from to	0 PSI 12 PSI
Maximum (red radial)		12 PSI
<u>Tachometer</u> :		
Recommended idle		650 RPM
Normal range (green arc)	to and from	500 RPM 2000 RPM 2350 RPM
Avoid continuous operation (red arc)	to from to and	2600 RPM 2000 RPM 2350 RPM 2600 RPM
Do Not exceed (red radial) Avoid continuous operation (re in aerobatic and full throttle	to ed arc) abov e level flig	2700 RPM 2700 RPM re 2600 RPM ht

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SECTION I

OPERATING LIMITATIONS

C. Weights

Maximum gross weight

1150 LBS.

NOTE: Reference station, FS 0.00 is located 60.56 inches forward of lower wing leading edge.

Maximum oil 2 U.S. Gals. 15 LBS: Fuel tank capacity 20 U.S. Gals. 120 LBS. Baggage, maximum 15 LBS. Usable fuel, normal flight 19 U.S. Gals. 114 LBS.

(See Section V, "Weight and Balance," for allowable weight and center of gravity combinations and detail loading instructions.)

Weight and Center of Gravity Limits:

Most forward limit:

FS 59.35 (17.53%MAC) at 1115 lbs. or less;

Most forward at maximum gross weight:

FS 60.37 (20.47%MAC) at 1150 lbs.;

Most rearward at maximum gross weight:

FS 61.38 (23.39%MAC) at 1150 lbs.;

Most rearward limit:

FS 62.48 (26.6% MAC) at 970 lbs. or less; with straight line variation between points given.

D. Flight Load Factors:

Positive flight, limit +6.0 G. Negative flight, limit -4.67 G.

Maneuvers and entry speeds:
See section IV, "Placards."

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SECTION I

OPERATING LIMITATIONS

Ε. Flight Limitations:

This airplane must be operated as a day VFR airplane only.

Flight into know icing conditions is prohibited.

No acrobatic maneuvers with baggage.

F. <u>Usable Fuel:</u>

Of the 20 U.S. gallon fuel tank capacity, 19 gallons are usable during all normal flight conditions. Unusable fuel, normal flight: 1 U.S. Gallon.

PITTS AEROBATICS AIRPLANE FLIGHT MANUAL MODEL S-1T AIRPLANE

SECTION II

OPERATING PROCEDURES

Normal Procedures:

- Opening Canopy: a.
- 1. To open canopy from outside, pull up on the canopy latch tabs located at the lower front edges of the canopy, and slide the canopy aft.
- 2. To open canopy from inside, pull aft on cable latch release located at top forward center of canopy.
- Check Stall Warning System as follows before every flight if aircraft does not have a full electrical system:
 - Reach into cockpit and turn master stall warn switch ON.
 - Walk to stall warn sensor on right wing and deflect vane UP.
 - If horn is clearly audible from wing, system is acceptable for flight. If horn is not clearly audible from wing, replace dry-cell battery before flight.
- c. Starting Engine from Cold:

	Alternate air:	OFF
2.	Propeller governor control:	HIGH RPM
3.	Fuel selector:	ON
4.	Master stall warn switch	ON
5.	Mixture control:	FULL RICH
6.	Throttle:	OPEN ½ INCH
7.	Auxiliary fuel pump:	PUMP 3 STROKES
	Pull mixture control to:	IDLE CUT-OFF
9.	Switch ignition to:	LEFT MAGNETO
10.		
11.		control
	immediately to:	FULL RICH
12.		BOTH MAGNETOS
	Oil pressure:	INDICATION
	•	

NOTE: Hot idle oil pressure 25 PSI minimum.

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SECTION II

OPERATING PROCEDURES

Α.	Normal	Procedures	(cont'd):

d.	Starting	Engine	From	Hot:

1	Alternate air:	
<u>.</u>	niccinate all:	OFF
2.	Propeller governor control;	HIGH RPM
3.	Fuel selector:	ON
4.	Master stall warn switch:	
	Minter 1	ON
٦.	Mixture control;	FULL RICH
6.	Throttle:	HALF OPEN
7.	Auxiliary fuel pump:	
	J - war p camp .	PUMP 1 STROKE
	Throttle	OPEN ½ INCH
9.	Pull mixture control to:	
10	Crital danie	IDLE CUT-OFF
TU.	Switch ignition to;	LEFT MAGNETO
11.	Prop engine.	1 121011210
12	Whon ongine stants - 1	_
12.	When engine starts, push mixture con	ntrol

12 immediately to: FULL RICH

13. Switch ignition to:
14. Oil pressure FULL KICH BOTH MAGNETOS 14. Oil pressure INDICATION

Ground Running and Warm-Up: e.

To prevent overheating follow these procedures:

1. Head airplane into wind

2. Mixture: Mixture: FULL RICH Propeller governor control HIGH RPM Warm-up at approximately: 1000 - 1200 RPM 3.

Avoid prolonged idling and do not exceed: 2200 RPM

f. <u>Pre-Takeoff</u>:

1. Warm-up as above.

2. Oil pressure: 3. Oil temperature: 4. Mixture control:

5. Elevator trim: 6. Flight controls:

Fuel pressure:

GREEN ARC

GREEN ARC FULL RICH NEUTRAL

FREE GREEN ARC

Set throttle to 1700 RPM and move propeller governor control through full range and return to: HIGH RPM

SECTION II

OPERATING PROCEDURES

A. Normal Procedures (cont'd):

- f. Pre-Takeoff: (cont'd)
- 9. Magneto check:

With propeller set at high RPM set throttle at:

2200 RPM

10. Switch magnetos from both to one and note drop off, return to both until engine regains speed and switch to other magneto and note drop-off,

then return to both.
Normal drop-off is:
Maximum drop-off is:

100 RPM

175 RPM

Difference in drop-off between magnetos is: 50 RPM

g. Landing:

1. Mixture control:

FULL RICH

2. Propeller governor control:

HIGH RPM

- h. Engine Shut-down:
 - 1. Throttle:

CLOSED

2. Mixture control:

IDLE CUT-OFF

3. Master stall warn switch:

OFF

4. Ignition switch:

OFF

i. Acrobatic Flight:

Low altitude acrobatics with less than $\frac{1}{4}$ tank of fuel onboard is not recommended.

j. Spin Recovery Procedure:

With aileron neutral, apply full opposite rudder briskly, followed by full nose down elevator. When spin rotation stops, neutralize rudder and elevator and recover to level flight.

NOTE: Aileron held against the spin may delay the

recovery.

B. <u>Emergency Procedures:</u>

- a. <u>In-flight Engine Restart:</u>
 - 1. Pull mixture control to:

IDLE CUT-OFF

2. Establish glide at:

100 MPH IAS

SECTION II

OPERATING PROCEDURES

- Normal Procedures (cont'd):
 - In-Flight Engine Restart (cont'd): a.
 - 3. Fuel selector:

ON ON

4. Master stall warn switch:

BOTH MAGNETOS

5. Ignition switch: 6. Throttle:

七 OPEN

7. Propeller governor control:

HIGH RPM

- Increase airspeed to start propeller wind-milling, if it has stopped.
- Advance mixture control to:

FULL RICH

b. Freezing of Pitot-Static Head:

In the event of icing of the static orifices on the pitot static head, an alternate source of static pressure is provided. To open the alternate static pressure source, turn the indicated valve on the instrument panel counterclockwise to full open. See placard for altitude error.

c. Best Glide Speed, Engine Out Is:

97 MPH IAS

- In Case of Emergency Bailout, pull canopy full aft prior to bailing out.
- NOTE: Stall warning is inoperative with master stall warn switch "OFF"

SECTION III

PERFORMANCE INFORMATION

A. Altitude loss during power off stalls:

200 FT

Power-off stalling speed versus bank angle, at 1150 lbs. gross weight and forward gross C.G. =

BANK ANGLE	STALLING SPEED	BANK ANGLE	STALLING SPEED
0°	64 MPH CAS	45°	76 MPH CAS
30°	69 MPH CAS	60°	91 MPH CAS

C. Demonstrated flight time, inverted is: 3 MINUTES

D. Demonstrated crosswind velocity is:

20 MPH

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SECTION IV

PLACARDS

The following placards are installed in the airplane:

- 2. Adjacent to airspeed indicator: "Design maneuver speed 154 MPH: demonstrated crosswind velocity 20 MPH."
- 3. On inside of baggage compartment door: "No acrobatics with baggage; max. baggage 15 lbs."
- 4. Adjacent to fuel tank filler neck: "Fuel 100/130 octane.
 19 gals. usable."
- 5. On instrument panel adjacent to alternate static source valve: "Open for alternate static"."
- 6. On left side of cockpit, adjacent to mixture control: "Pull for lean mixture."
- 7. On right side of cockpit, adjacent to alternate engine inlet air control: "Pull for alternate air."
- 8. On left side of cockpit, adjacent to elevator trim control handle: "Nose up; Neutral; Nose down."
- 9. On throttle quadrant: "Open; Throttle; Closed."
- 10. On instrument panel: "No Smoking."
- 11. Adjacent to master stall warn switch: "Master stall warn.;
 Must be on for flight; ON; OFF."
- 12. Adjacent to propeller governor control on LH side of cockpit: "Push for high RPM."
- 13. On LH side of cockpit fairing, inside in clear view of pilot:

 "This airplane must be operated as an aerobatic category
 airplane in compliance with the operating limitations stated
 in the form of placards markings and manuals. Operations
 limited to day VFR conditions. Flight into known icing
 conditions prohibited."

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SECTION IV

PLACARDS

13. cont'd

"APPROVED MANEUVERS AND RECOMMENDED ENTRY SPEEDS: (MPH)

MANEUVER	M.A	INSIDE AX. MIN.	O MAX	rside . Min.
LOOP (UP) LOOP (DOWN) SLOW ROLL BARREL ROLL SNAP ROLL HAMMERHEAD LAZY EIGHT CHANDELLE STALLS AND SPINS	18 10 18 18 14 18 18	70 70 100 100 130 0 90 0 130 0 140	180 100 180 180 110 180 180 180 J DECELERAT	130 70 100 130 90 130 140 140 10N)

For spin recover, put ailerons neutral, apply full opposite rudder briskly and then apply nose down elevetor."

- 14. On instrument panel, adjacent to "F" mark on fuel quantity indicator: "19 gals. usable."
- 15. On LH side of cockpit fairing in clear view of pilot: "No acrobatics with baggage."
- 16. On top centerline of canopy at forward edge: "Pull aft to open."
- 17. On right hand forward lower corner of canopy: "Do not open past first notch above 120 MPH IAS."
- 18. On instrument panel adjacent to alternate static source valve:

Alternate		On:	Airspeed,	<u>MPH</u>	Error, Ft.
	•		80		-20
			100		-10
			120		+60
			140		+90
			160		+150

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SECTION IV

PLACARDS (cont'd)

21. On outside lower forward corners of canopy bubble (both sides): "To open: Lift tab slide aft (both sides)."